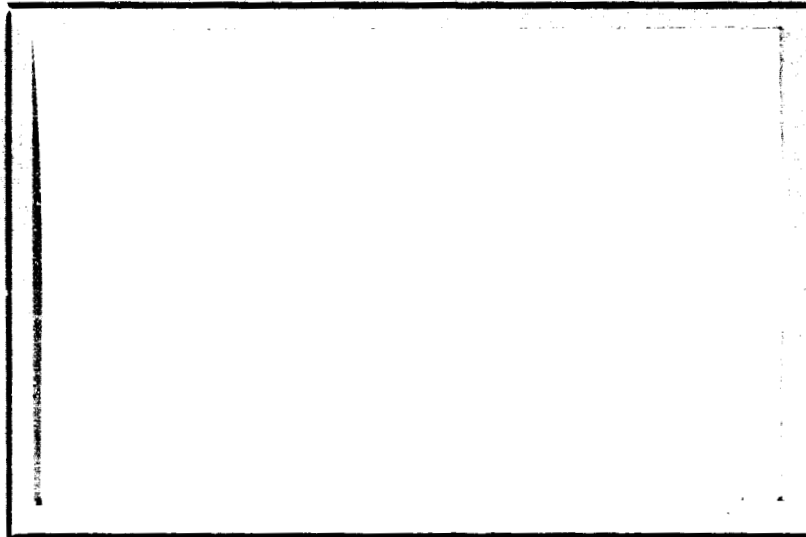


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OF THE ANNULAR SUSPENSION POINTING SYSTEM,
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FUNCTIONAL SPECIFICATIONS OF THE
ANNULAR SUSPENSION POINTING SYSTEM

by

Bryan Edwards
Department of Computer Science
University of Colorado at Boulder
Boulder, Colorado

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Abstract

The following is a description of the Annular Suspension Pointing System. This description is written using the Design Realization, Evaluation and Modelling (DREAM) system, and its design description technique, the DREAM Design Notation (DDN).

Appendix A contains a DDN description of the Annular Suspension Pointing System. The information contained in this description was derived from the NASA-produced report, "The Executive Software For the Annular Suspension Pointing System," which appears as Appendix B. The description is divided into four major sections.

The first section of Appendix A (System Overview) contains the major units of the system, their interconnections, and the event flow between these units. Figure 1 corresponds to Figure 1 in the original report, with the addition of three major units: analog sources, experiment computer, and the system operator. Additional communication paths are also shown. Each communications path is labeled with a number. These numbers correspond to the CONNECTIONS given in the DREAM description. In addition, the EVENT DEFINITIONS reference the communications paths which the events use, by appending the path number to the event name. Finally, the legal event sequences are given in the DESIRED BEHAVIOR section of the description, using a regular expression type notation. In this section, a shorthand, non-standard, notation is used to indicate the repetition of a sequence of events a specific number of times.

The second section (LEVEL II) describes the basic operations of each of the major units of the system. The input and output ports are identified, and an abstract model of the operation is given in terms of the input and output.

In the third section (LEVEL III), the notion of the internal servicers P(1), P(2) and P(3) is introduced. The internal operation of these servicers is not detailed. The logical interaction between the servicers and the input and output ports of the NASA standard space computer is given.

The notion of the time intervals T(1), T(2) and T(3) is introduced in the fourth section (LEVEL IV). Here we see the interaction between the master timing pulse and the signals to the three processes P(1), P(2) and P(3).

In Appendix B, we have included a copy of the NASA-produced functional specification of the Annular Suspension Pointing System.

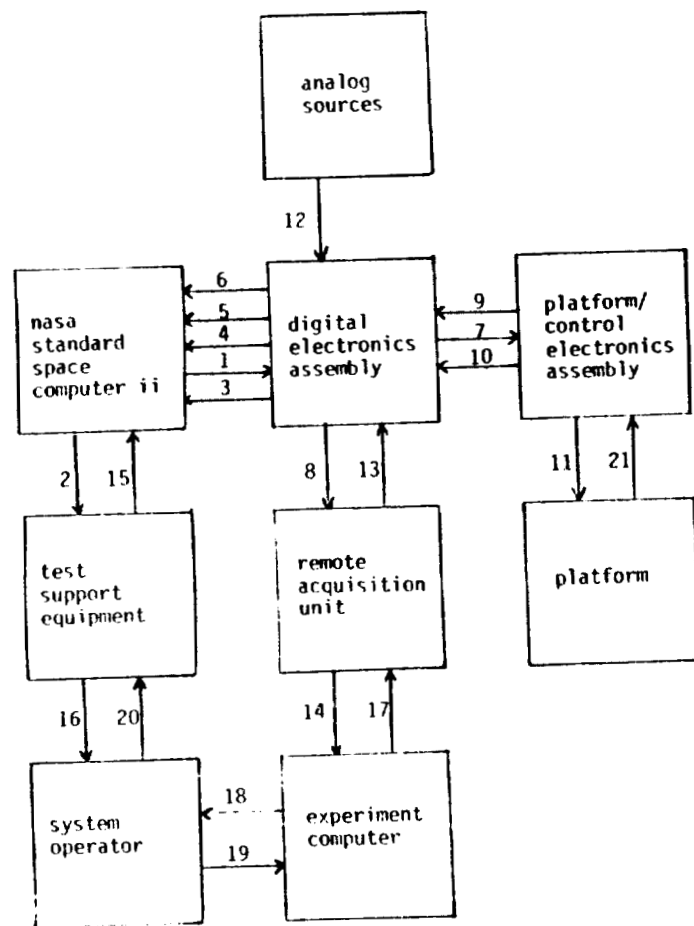
The portions of the report which were captured by the DDN description of the system are underlined.

Some portions of the NASA report contain very detailed descriptions of sections of the system. This detail is not reflected in the DDN description. Further elaborations of the DDN description would be required to capture this detail.

The DDN description does not capture the notion of the mode (idle, coarse, fine, slew) of the system, and the details of the data communicated between the system units is not given. The NASA report does not contain enough information in these areas to allow further elaboration.

In preparing the DDN description, the concepts available in DDN were adequate to describe most of this embedded computer system. The only area that DDN does not adequately describe is the notion of a specific interval of time.

FIGURE 1
Communications paths between system components



Appendix A:
DREAM Design Notation Description
of
Annular Suspension Pointing System

SYSTEM OVERVIEW

[annular_suspension_pointing_system]: SUBSYSTEM CLASS;

DOCUMENTATION;

The purpose of the ASPS is to control a platform which will be flown on the space shuttle. Equipment (e.g., a telescope) will be mounted on the platform and the ASPS will allow this equipment to be pointed in a given direction with extreme accuracy ($\pm 4.84 \times 10^{-7}$ radians) and this position maintained for extended periods (stability $\pm 4.84 \times 10^{-8}$ radians /sec) in the presence of shuttle disturbances.

END DOCUMENTATION;

QUALIFIERS;

t1_per_t2, t2_per_t3

END QUALIFIERS;

SUBCOMPONENTS;

nsscii OF [nasa_standard_space_computer_ii],
 dea OF [digital_electronics_assembly],
 tse OF [test_support_equipment],
 rau OF [remote_acquisition_unit],
 ec OF [experiment_computer],
 pea_cea OF [platform_electronics_assembly_control_electronics_assembly],
 p OF [platform],
 as OF [analog_sources],
 so OF [system_operator]

END SUBCOMPONENTS;

CONNECTIONS;

PLUG (nsscii|dea_outputs, dea|nsscii_inputs), 1
 PLUG (nsscii|tse_outputs, tse|nsscii_inputs), 2
 PLUG (dea|nsscii_outputs, nsscii|dea_inputs), 3
 PLUG (dea|t1_tick, nsscii|t1_tick), 4
 PLUG (dea|t2_tick, nsscii|t2_tick), 5

SYSTEM OVERVIEW

PLUG (dea|t3_tick, nsscii|t3_tick), 6
 PLUG (dea|pea_cea_outputs, pea_cea|dea_inputs), 7
 PLUG (dea|rau_outputs, rau|dea_inputs), 8
 PLUG (pea_cea|master_timing_pulse, dea|master_timing_pulse), 9
 PLUG (pea_cea|dea_outputs, dea|pea_cea_inputs), 10
 PLUG (pea_cea|p_outputs, p|pea_cea_inputs), 11
 PLUG (as|dea_outputs, dea|as_inputs), 12
 PLUG (rau|dea_outputs, dea|rau_inputs), 13
 PLUG (rau|ec_outputs, ec|rau_inputs), 14
 PLUG (tse|nsscii_outputs, nsscii|tse_inputs), 15
 PLUG (tse|so_outputs, so|tse_inputs), 16
 PLUG (ec|rau_outputs, rau|ec_inputs), 17
 PLUG (ec|so_outputs, so|ec_inputs), 18
 PLUG (so|ec_outputs, ec|so_inputs), 19
 PLUG (so|tse_outputs, tse|so_inputs), 20
 PLUG (p|pea_cea_outputs, pea_cea|p_inputs), 21

END CONNECTIONS;

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SYSTEM OVERVIEW

[asps_operation]: EVENT CLASS;

EVENT DEFINITION;

system_operator_request_experiment_19: DESCRIPTION;

This event corresponds to the system operator entering a request, at the operator console, to the experiment computer.

END DESCRIPTION;

experiment_computer_request_platform_action_17,13: DESCRIPTION;

In order to perform a given experiment, the experiment computer must manipulate the platform in some predefined manner.

END DESCRIPTION;

dea_request_computation_3: DESCRIPTION;

Many times, computations must be performed before a requested platform action can occur.

END DESCRIPTION;

nsscii_computation_result_returned_1: DESCRIPTION;

Computations supporting the platform are performed in the nsscii. Results are returned to the dea.

END DESCRIPTION;

dea_request_platform_action_7,11: DESCRIPTION;

The platform is actually controlled by the pea/cea.

END DESCRIPTION;

platform_responds_21,10: DESCRIPTION;

The platform responds to request from the pea/cea.

END DESCRIPTION;

platform_result_returned_to_experiment_computer_8,14: DESCRIPTION;

The results of a high-level platform operation are returned to the experiment computer.

END DESCRIPTION;

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SYSTEM OVERVIEW

experiment_result_returned_to_system_operator_18: DESCRIPTION;

The result of our experiment is returned to the system operator, at the operator console.

END DESCRIPTION;

system_operator_request_test_20: DESCRIPTION;

This event corresponds to the system operator entering a request at the test console, to the test support equipment.

END DESCRIPTION;

tse_request_action_15: DESCRIPTION;

In order to perform a given test, the test support equipment must get certain data from the nsscii.

END DESCRIPTION;

nsscii_result_returned_2: DESCRIPTION;

Test data from the nsscii is returned to the test support equipment.

END DESCRIPTION;

test_result_returned_to_system_operator_16: DESCRIPTION;

The result of a test is returned to the system operator, at the test console.

END DESCRIPTION;

master_timing_pulse_9: DESCRIPTION;

This pulse is generated every T(1) milliseconds.

END DESCRIPTION;

t1_timing_pulse_4: DESCRIPTION;

This pulse is generated every T(1) milliseconds as a result of the master timing pulse.

END DESCRIPTION;

t2_timing_pulse_5: DESCRIPTION;

This pulse is generated every T(2) milliseconds as a result of the master timing pulse. Note that T(2) milliseconds is an integral multiple of T(1).

END DESCRIPTION;

SYSTEM OVERVIEW

t3_timing_pulse_6: DESCRIPTION;

This pulse is generated every T(3) milliseconds as a result of the master timing pulse. Note that T(3) milliseconds is an integral multiple of T(2).

END DESCRIPTION;

SYSTEM OVERVIEW

DESIRED BEHAVIOR;

```
SHUFFLE(
  REPEAT(
    SEQUENCE(
      SEQUENCE(
        SEQUENCE(master_timing_pulse_9, t1_timing_pulse_4), t1_per_t2
        t2_timing_pulse_5), t2_per_t3
        t3_timing_pulse_6)),
    REPEAT(
      SEQUENCE(
        system_operator_request_experiment_19,
        REPEAT(
          SEQUENCE(
            experiment_computer_requests_platform_action_17,13,
            REPEAT
              (SEQUENCE
                (dea_request_computation_3,
                 nsscii_computation_result_returned_1)),
            REPEAT
              (SEQUENCE
                (dea_request_platform_action_7,11,
                 platform_responds_21,10)),
                platform_result_returned_to_experiment_computer_8,14)),
            experiment_result_returned_to_system_operator_18)),
        REPEAT(
          SEQUENCE(
            system_operator_request_test_20,
            REPEAT(
              SEQUENCE(
                tse_request_action_15,
                nsscii_results_returned_2)),
            test_result_returned_to_system_operator_16)))
```

END DESIRED BEHAVIOR;

END EVENT DEFINITION;

END EVENT CLASS;

END SUBSYSTEM CLASS;

```
[system_operator]: SUBSYSTEM CLASS;

ec_outputs: OUT PORT;
END PORT;

tse_outputs: OUT PORT;
END PORT;

ec_inputs: IN PORT;
END PORT;

tse_inputs: IN PORT;
END PORT;

operator: CONTROL PROCESS;

MODEL;

ITERATE

SELECT
(PERHAPS): SEND ec_outputs;
(PERHAPS): SEND tse_outputs;
END SELECT;

SELECT
(PERHAPS): RECEIVE ec_inputs;
(PERHAPS): RECEIVE tse_inputs;
END SELECT;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;
```

```
[experiment_computer]: SUBSYSTEM CLASS;

rau_outputs: OUT PORT;
END PORT;

so_outputs: OUT PORT;
END PORT;

rau_inputs: IN PORT;
END PORT;

so_inputs: IN PORT;
END PORT;

experiment: CONTROL PROCESS;

MODEL;

ITERATE

RECEIVE so_inputs;
SEND rau_outputs;
RECEIVE rau_inputs;
SEND so_outputs;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;
```

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LEVEL II

[remote_acquisition_unit]: SUBSYSTEM CLASS;

dea_outputs: OUT PORT;
END PORT;

ec_outputs: OUT PORT;
END PORT;

dea_inputs: IN PORT;
END PORT;

ec_inputs: IN PORT;
END PORT;

acquisition: CONTROL PROCESS;

MODEL;

ITERATE

RECEIVE ec_inputs;
SEND dea_outputs;
RECEIVE dea_inputs;
SEND ec_outputs;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

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LEVEL II

[platform]: SUBSYSTEM CLASS;

pea_cea_outputs: OUT PORT;
END PORT;

pea_cea_inputs: IN PORT;
END PORT;

platform: CONTROL PROCESS;

MODEL;

ITERATE

RECEIVE pea_cea_inputs;
SEND pea_cea_outputs;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

-15-

LEVEL II

[analog_sources]: SUBSYSTEM CLASS;

dea_outputs: OUT PORT;
END PORT;

source: CONTROL PROCESS;

MODEL;

ITERATE

SEND dea_outputs;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

-16-

LEVEL II

[test_support_equipment]: SUBSYSTEM CLASS;

nsscii_outputs: OUT PORT;
END PORT;

so_outputs: OUT PORT;
END PORT;

nsscii_inputs: IN PORT;
END PORT;

so_inputs: IN PORT;
END PORT;

support: CONTROL PROCESS;

MODEL;

ITERATE

RECEIVE so_inputs;
SEND nsscii_outputs;
RECEIVE nsscii_inputs;
SEND so_outputs;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

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LEVEL II

[platform_electronics_assembly_control_electronics_assembly]: SUBSYSTEM CLASS;

master_timing_pulse: OUT PORT;
END PORT;

dea_outputs: OUT PORT;
END PORT;

p_outputs: OUT PORT;
END PORT;

dea_inputs: IN PORT;
END PORT;

p_inputs: IN PORT;
END PORT;

control: CONTROL PROCESS;

MODEL;

ITERATE

SELECT

(PERHAPS): SEND master_timing_pulse;

(PERHAPS): RECEIVE dea_inputs;

SEND p_outputs;

RECEIVE p_inputs;

SEND dea_outputs;

END SELECT;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

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LEVEL II

[nasa_standard_space_computer_ii]: SUBSYSTEM CLASS;

dea_outputs: OUT PORT;
END PORT;

tse_outputs: OUT PORT;
END PORT;

dea_inputs: IN PORT;
END PORT;

t1_tick: IN PORT;
END PORT;

t2_tick: IN PORT;
END PORT;

t3_tick: IN PORT;
END PORT;

tse_inputs: IN PORT;
END PORT;

nsscii_executive: CONTROL PROCESS;

MODEL;

perform_initialization;

SEND dea_outputs;

RECEIVE dea_inputs;

ITERATE

SELECT

(PERHAPS): RECEIVE t1_tick;

SEND dea_outputs;

(PERHAPS): RECEIVE t2_tick;

(PERHAPS): RECEIVE t3_tick;

(PERHAPS): RECEIVE tse_inputs;

SEND tse_outputs;

END SELECT;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

-19-

LEVEL II

[digital_electronics_assembly]: SUBSYSTEM CLASS;

nssci outputs: OUT PORT;
END PORT;

t1 tick: OUT PORT;
END PORT;

t2 tick: OUT PORT;
END PORT;

t3 tick: OUT PORT;
END PORT;

pea cea outputs: OUT PORT;
END PORT;

rau outputs: OUT PORT;
END PORT;

nssci inputs: IN PORT;
END PORT;

master timing_pulse: IN PORT;
END PORT;

pea cea inputs: IN PORT;
END PORT;

as inputs: IN PORT;
END PORT;

rau inputs: IN PORT;
END PORT;

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LEVEL II

dea_executive: CONTROL PROCESS;

MODEL;

ITERATE

SELECT

(PERHAPS): RECEIVE master timing_pulse;
SEND t1_tick;

SELECT

(PERHAPS) SEND t2_tick;

(PERHAPS) SEND t2_tick;

SEND t3_tick;

END SELECT;

(PERHAPS): RECEIVE rau_inputs;

MAYBE

SEND nssci_outputs;

RECEIVE nssci_inputs;

END MAYBE;

MAYBE

SEND pea cea outputs;

RECEIVE pea cea inputs;

END MAYBE;

(PERHAPS): RECEIVE as_inputs;

END SELECT;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;

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LEVEL III

'[nasa_standard_space_computer_ii]: SUBSYSTEM CLASS'

SUBCOMPONENTS;

p1, p2, p3 OF [process]

END SUBCOMPONENTS;

CONNECTIONS;

PLUG (nssci_executive|p1_initiate, p1|initiate),
PLUG (nssci_executive|p2_initiate, p2|initiate),
PLUG (nssci_executive|p3_initiate, p3|initiate),
PLUG (p1|complete, nssci_executive|p1_complete),
PLUG (p2|complete, nssci_executive|p2_complete),
PLUG (p3|complete, nssci_executive|p3_complete)

END CONNECTIONS;

nssci_executive: CONTROL PROCESS;

p1_initiate: LOCAL OUT PORT;
END PORT;

p2_initiate: LOCAL OUT PORT;
END PORT;

p3_initiate: LOCAL OUT PORT;
END PORT;

p1_complete: LOCAL IN PORT;
END PORT;

p2_complete: LOCAL IN PORT;
END PORT;

p3_complete: LOCAL IN PORT;
END PORT;

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LEVEL III

MODEL;

perform initialization;
SEND dea_outputs;
RECEIVE dea_inputs;
ITERATE

SELECT

(PERHAPS): RECEIVE t1_tick;
SEND p1_initiate;
(PERHAPS): RECEIVE t2_tick;
SEND p2_initiate;
(PERHAPS): RECEIVE t3_tick;
SEND p3_initiate;
(PERHAPS): RECEIVE p1_complete;
SEND dea_outputs;
(PERHAPS): RECEIVE p2_compleie;

(PERHAPS): RECEIVE p3_complete;

(PERHAPS): RECEIVE tse_inputs;
SEND tse_outputs;

END SELECT;

END ITERATE;

END MODEL;

END CONTROL PROCESS;

LEVEL III

```
[process]: SUBSYSTEM CLASS;

  complete: OUT PORT;
  END PORT;

  initiate: IN PORT;
  END PORT;

  process: CONTROL PROCESS;

  MODEL;

    ITERATE
      RECEIVE initiate;
      perform operations;
      SEND complete;

    END ITERATE;

  END MODEL;

END CONTROL PROCESS;

END SUBSYSTEM CLASS;
```

LEVEL IV

'[digital_electronics_assembly]: SUBSYSTEM CLASS'

```
  QUALIFIERS;
    t1_per_t2, t2_per_t3

  END QUALIFIERS;

  LOCAL SUBCOMPONENT;

    t1_pulses OF [0:: t1_per_t2],
    t2_pulses OF [0:: t2_per_t3]

  END LOCAL SUBCOMPONENT;

  dea_executive: CONTROL PROCESS;

  MODEL;

    SET t1_pulses TO 0;
    SET t2_pulses TO 0;

    ITERATE

      SELECT
        (PERHAPS): RECEIVE master_timing_pulse;
        SEND t1_tick;
        SET t1_pulses TO t1_pulses + 1;
        IF t1_pulses = t1_per_t2 THEN
          SET t1_pulses TO 0;
          SEND t2_tick;
          SET t2_pulses TO t2_pulses + 1;
          IF t2_pulses = t2_per_t3 THEN
            SET t2_pulses TO 0;
            SEND t3_tick;
          END IF;
        END IF;

        (PERHAPS): RECEIVE rau_inputs;
        MAYBE
          SEND nscii_outputs;
          RECEIVE nscii_inputs;
        END MAYBE
        MAYBE
          SEND pea_cea_outputs;
          RECEIVE pea_cea_inputs;
        END MAYBE;

        (PERHAPS): RECEIVE as_inputs;

      END SELECT;

    END ITERATE;

  END MODEL;

END CONTROL PROCESS;
```


Appendix B:
Functional Requirements
of
Annular Suspension Pointing System

(Produced by NASA
Langley Research
Center Personnel)

THE EXECUTIVE SOFTWARE FOR THE ANNULAR SUSPENSION POINTING SYSTEM

1. INTRODUCTION.

This document attempts to describe the Annular Suspension Pointing System (ASPS) hardware facilities and the structure of the software executive in sufficient detail that it can be used as an example of the requirements for concurrent programming in NASA embedded computer systems. The hardware details are provided for those who are unfamiliar with the general layout of the ASPS. This description is intended to be accurate and every effort will be made to ensure that it correctly reflects the software currently being written for the ASPS engineering model. The engineering model is a ground-based system used for testing. This is the first version of the software which will be used. This description is also intended to be complete in the sense that the functions of the software is defined in sufficient detail (albeit informally) that only minor parametric details are needed before the software can be constructed.

Two consequences of the fact that the software described is for an engineering model are that the software is instrumented and the existence of a human operator is assumed. The instrumentation is for performance evaluation and error analysis. It will not be specified here since it does not affect the ASPS executive function.

The purpose of the ASPS is to control a platform which will be flown on the Space Shuttle. Equipment (e.g. a telescope) will be mounted on the platform and the ASPS will allow this equipment to be pointed in a given direction with extreme accuracy (\pm or $- 4.84 \times 10^{-7}$ radians) and this position maintained for extended periods (stability \pm or $- 4.84 \times 10^{-8}$ radians per sec.) in the presence of Shuttle disturbances.

II. HARDWARE CONFIGURATION OF THE SYSTEM

Figure 1 shows the organization of the major hardware units comprising the ASPS.

Definitions:

a) TSE: Test Support Equipment. The TSE consists of a terminal and a computer. It will be used to generate various inputs from the operator and display messages to the operator for ground testing of the ASPS.

b) NSSC II: NASA Standard Space Computer II. The NSSC II is basically an IBM System 360 computer. It will be used to carry out the computations which implement the control laws for the platform.

c) DEA: Digital Electronics Assembly. The DEA is an electronics assembly based on a 2-80 microprocessor which is used as an I/O controller for the NSSC II. Figure 2 shows the major hardware units of the DEA.

d) PEA/CLA: Platform Electronics Assembly/Control Electronics Assembly. The PEA is the electronic assembly on the platform which, together with the CLA, is responsible for moving the platform and sending position information to the NSSC II.

e) RAU: Remote Acquisition Unit. The RAU provides 32 digital input and 32 digital output lines. It is connected to the experiment computer i.e. the experiment which is using the ASPS, and all I/O between the DEA and the experiment computer is through the RAU.

The system of interest for the executive consists of an NSSC II computer connected to the DEA. The DEA is connected to the NSSC II by 16 inputs, 16 output and several control lines. During flight all I/O to the NSSC II is through the DEA. In the engineering model, I/O to

the NSSC II can also be from the TSE. Analog data to or from the platform is converted (A/D, D/A) in the DEA and preprocessed in the DEA before being sent to the NSSC II.

The experiment computer sends platform control commands (e.g. point telescope in a particular direction) to the DEA, which in turn requests the NSSC II to compute the control laws. This output is sent through the DEA to the PEA/CLA which moves the platform.

The system master timing pulse is generated by the PLA and sent to the DEA. This is a pulse every T(1) milliseconds which is used for real-time timing. T(1) is a fixed integer whose value has not been finalized.

NOTE: This pulse is not sent directly to the NSSC II.

III. NSSC II CHARACTERISTICS

This section provides a summary only and is not intended to be complete. Full details of the machine can be obtained from the hardware reference manual.

The NASA Standard Space Computer II (NSSC II) is very much like an IBM System 360. A thorough knowledge of S/360 is assumed in this summary. The NSSC II's instruction set contains 83 of the 87 instructions from the S/360 Standard Set. (Recall that the Standard Set does not include decimal, direct control, protection or floating point instructions.) The exceptions are HIO(9E), SIO(9C), TCH(9F), and TIO(9D). The semantics of these 83 instructions are identical to S/360 except for the following areas:

(1) The S/360 interval timer at location 80(decimal) is not implemented.

(2) Effective addresses are limited to 16 bits except for the LA(41) instruction which generates a 24-bit result.

Added to these 83 instructions are three new instructions:

	Mnemonic	Opcode	Format
Timer Read and Set	THRS	A4	RS
Start I/O	SIO	A5	RS
Set Storage Key	SSE	08	RR

The NSSC II supports a real-time clock and an interval timer. The real-time clock is 32 bits long, is incremented by 1 each 112.64 microseconds and causes no interrupt on overflow. The interval timer is 16 bits long, is decremented by 1 each 112.64 microseconds and generates an external interrupt on change of sign from positive to negative.

The clocks are read, or read and set, individually by the THRS instruction. Reading yields a timer value in a register. Setting involves a value from storage being placed in the timer.

There are four kinds of I/O. They are:

(a) Direct.

(b) Buffered.

(c) Direct Memory Access (DMA).

(d) External Interrupt.

Direct I/O constitutes transferring 16 bits of data to/from the NSSC II from/to the DEA via a 16 bit bus. Transfers of this type are the result of the NSSC II executing a SIO instruction. Note that this is totally different from the SIO instruction on S/360.

Buffered I/O is a means of performing block transfers of data to/from the NSSC II memory in parallel with normal execution of the NSSC II. When buffered I/O takes place, memory references and sequencing are controlled by hardware within the CPU but this does not interfere with instruction execution. The NSSC II has provision for up to 16 devices to perform buffered I/O. A fixed storage location is used on the NSSC II to point to a buffered I/O device table with 16 entries. An entry contains two words each of which is a word count and address pair. One word is for input and the other is for output. The word count is the number of words to be transferred and the address is the main memory buffer location. If the relevant word count is positive when a buffered I/O operation begins, then the count is decremented and the address is incremented in the table entry as each word is transferred. When the transfer is complete the word count will be zero (assuming no error). If the word count is initially negative, the word count is modified during the buffered I/O operation but is reset to its original value when the operation completes. The number of words transferred in this case is the absolute value of the word count. A buffered I/O operation is initiated using direct I/O (SIO instruction) to send 16

bits of data to the device which will perform the operation.

Direct memory access is literally direct access of the NSSC II memory. The ASPS system does not use DMA.

An external interrupt changes the state of the NSSC II as a result of an external stimulus and as such can be regarded as an input mechanism. The ASPS system does not use external interrupts for data input.

Memory on the NSSC II is protected in blocks of 1024 bytes. Storage keys are two bits long, and they are used for write protection only. One bit is used to inhibit CPU and buffered I/O storing and the other is used to inhibit DMA storing. Storage keys are set by the Set Storage Key (SSK) instruction, and also, following an interrupt, the storage key of the first block of memory is set to allow CPU and Buffered I/O storing but inhibit DMA storing. No other storage keys are affected by interrupts.

As well as the above, the NSSC II is equipped with a set of short precision (16-bit) instructions which operate with 16-bit fixed point two's complement numbers. They are manipulated in the lower half of the general-purpose registers and there is no sign extension as on a S/360. A long precision fixed point (64-bit) instruction set is available also. An even-odd register pair is used for holding 64-bit numbers and only ADD, SUBTRACT, COMPARE, LOAD and STORE instructions are provided.

IV. ASPS EXECUTIVE FUNCTIONAL DESCRIPTION.

The ASPS executive has the primary goal of providing scheduling in real time of certain processes. There are three time periods of interest. At present they are 10 milliseconds, 100 milliseconds and 1 second but these may be adjusted. These time periods will be referred to here by the symbols $T(1)$, $T(2)$, and $T(3)$, in millisecond time units. Associated with these time periods are three sets of processes. The set $\{P(1,j)\}$ is associated with $T(1)$, the set $\{P(2,j)\}$ with $T(2)$ and the set $\{P(3,j)\}$ with $T(3)$. Certain computations must be completed every $T(1)$ milliseconds, others every $T(2)$ milliseconds and still others every $T(3)$ milliseconds. $T(2)$ and $T(3)$ are integer multiples of $T(1)$, and $T(3)$ is an integer multiple of $T(2)$.

Timing is centered around an I/O interrupt from the DMA which is derived from, but not coincident with, the system master timing pulse. This interrupt will be generated by the DMA every $T(1)$ milliseconds regardless of what the NSSC II does, although the NSSC II can mask it. On system start-up, the ASPS executive performs any data initializations which are necessary, signals the DMA that initialization is complete using direct I/O, and then places the NSSC II in the wait state with no processes active.

Processing begins when the first I/O interrupt arrives from the DMA. From then on, $P(1,j)$ must be completed every $T(1)$, $P(2,j)$ every $T(2)$ and $P(3,j)$ every $T(3)$ milliseconds of real time for some j . Real time can be thought of as a sequence of $T(3)$ time periods. Each $T(3)$ time period is broken into an integral number of $T(2)$ time periods, each $T(2)$ time period is broken into an integral number of $T(1)$ time periods. The quantity j is called the mode and a different process is used for each mode. The mode is the operating state of the platform and currently n (i.e. the number of modes) is 4. They are called IDLE, COURSE, FINE and SLEW. The system changes mode based on certain inputs (see below) and only certain transitions are valid. Mode changes can only occur at the beginning of a $T(3)$ millisecond time period.

The NSSC II interval timer is not used for any determination of real time. It is used solely as a check on the system master timing pulse. At the beginning of each $T(1)$ time period (i.e. following the interrupt from the DMA) the interval timer is loaded with a value slightly larger than $T(1)$. If the timer ever expires then clearly an error has occurred. For the initial version of the executive, if the timer interrupt ever occurs, the system will not attempt to recover but merely inform the operator and enter the wait state.

When the I/O interrupt at the beginning of $T(1)$ occurs the executive is entered. The DMA will already have completed a buffered input operation and placed a total of $L(IN)$ words into the NSSC II memory. $L(IN)$ is currently 38. This block of data is in two parts. The first

L(1K)-2 words are data for process P(1,j) and the last 2 are input to another process (see below). Prior to initiating P(1,j), these two words are removed by the executive and used to build a table in a separate memory area.

When P(1,j) completes, a table of outputs of length L(OUT) have been produced. L(OUT) is currently 32. A direct output is sent by the executive to the DEA which then begins a buffered output operation, i.e. the DEA removes the results of P(1,j) for its own use.

After the direct output has been sent, part of P(2,j) is run. For every j, the process P(2,j) must be completed in a T(2) time period. It is organized as a series of subprocesses which, when executed in series constitute the entire process P(2,j). These subprocesses will be denoted P(2,j,k). For every j and k, the process P(1,j) and P(2,j,k) can be executed sequentially in less than T(1) milliseconds. Clearly k has to be less than or equal to T(2)/T(1) in order to meet the deadline. The breaking of P(2,j) into a series of subprocesses is not a requirement but merely the process structure in the present design.

When P(2,j,k) completes, P(3,j) is resumed. It continues to execute until either:

(a) the next I/O interrupt from the DEA occurs or

(b) P(3,j) completes.

P(3,j) must be completed in a T(3) time period. P(3,j) for the current j is initiated at the beginning of each T(3) time period and following the completion of P(3,j) the processor will be in the wait state if P(1,j) or P(2,j) are not executing. P(3,j) operates on a table of data which is constructed for execution i of P(3,j) during execution i-1. The table is L(BACK) words long and is in fact constructed from the 2 word blocks which were not input to P(1,j) during the executions of P(1,j) (see above). The first of these two words is a key and the second is a data word. If the key is negative, the data word is to be ignored. If the key is positive it consists of two parts. The first is an index indicating where in the data table the data word belongs. The second part is an identifier indicating to which of several possible tables the data word belongs. During any given T(3) time period all of the data words will be intended for the same data table. If the identifier changes during a given T(3) time period, an error has occurred.

Switching of modes can only occur between execution of P(3,j), i.e. at most only every T(3) milliseconds. One of the constituents of the data table for P(3,j) is a mode change indication. This designates the mode which the system will be in for the next T(3) time period. Valid mode

transitions have not yet been decided.

In addition to real time management, the executive must respond to the other sources of interrupt on the BSSC II. The machine check and program interrupts are both to be regarded as errors, and processing will consist of informing the operator and putting the system into the wait state. Supervisor call interrupts must provide supervisor services in the normal way and only two such services are presently defined. They are:

(1) SVC code 55(hex) - Process P(3,j) has ended.

(2) SVC code AA(hex) - Process P(1,j) has ended.

External interrupts are to be regarded as errors except for the interrupt generated by the operator from the TSL. Processing in this case is currently undefined and so all external interrupt processing consists of informing the operator and putting the system into the wait state.

7. SUMMARY OF TIMING.

Refer to Fig. 3 for the system timing:

- (a) The PEA generates a sync pulse every T(1) milliseconds.
- (b) Starting at the trailing edge of the sync pulse, the DEA
- (c) The DEA interrupts the NSSC II when input is complete.
- (d) The NSSC II computes with the data and deposits the output in the buffer.
- (e) The NSSC II then signals the DEA to indicate that data is available.
- (f) The DEA begins to remove data from the output buffer in the NSSC II.
- (g) The NSSC II then performs the next sequential parts of the T(2) msec computation.
- (h) Once (g) is complete the NSSC II reverts to background processing.
- (i) The sequence repeats.

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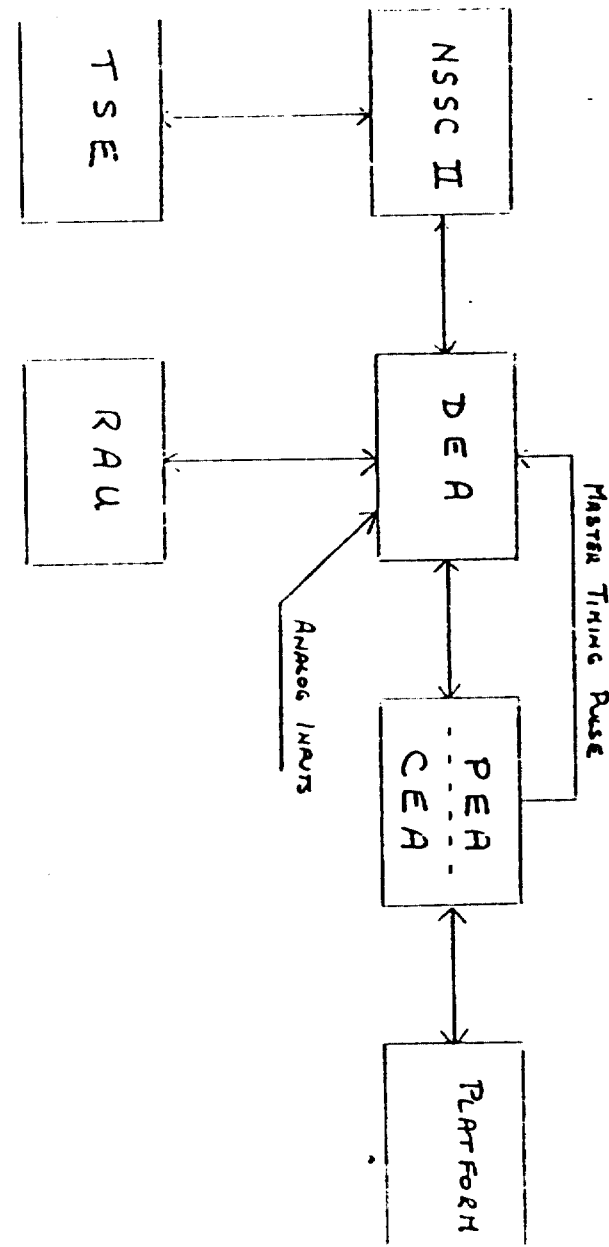


FIGURE 1

FIGURE 2

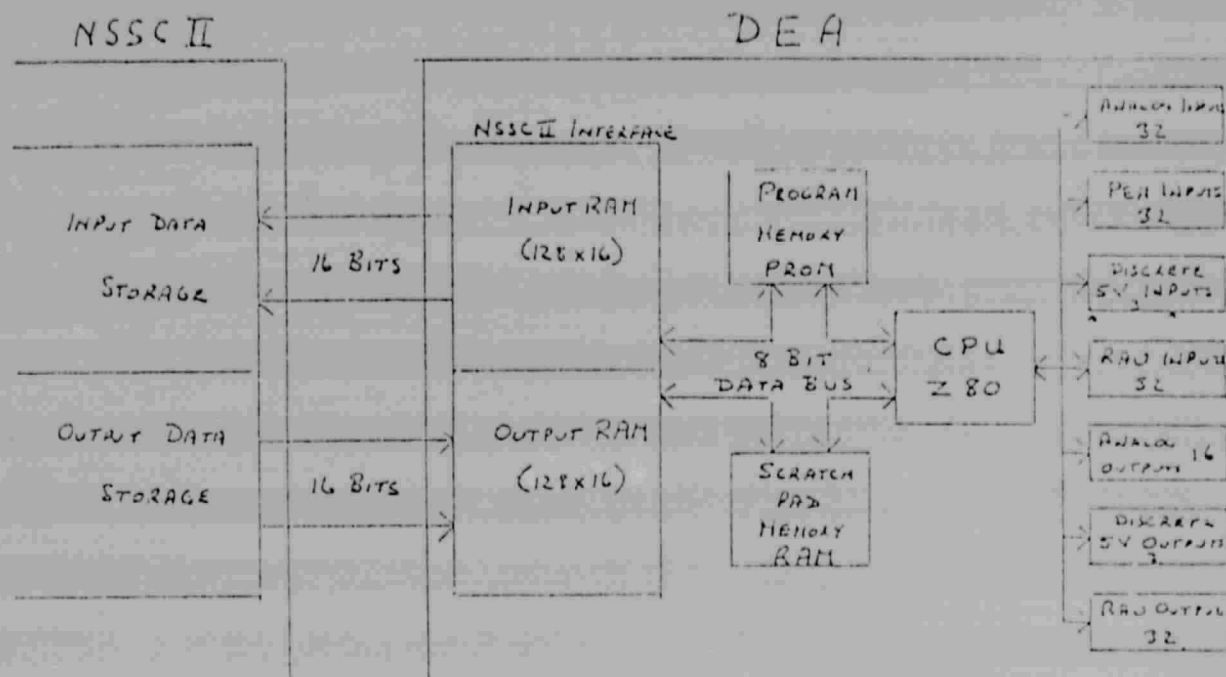
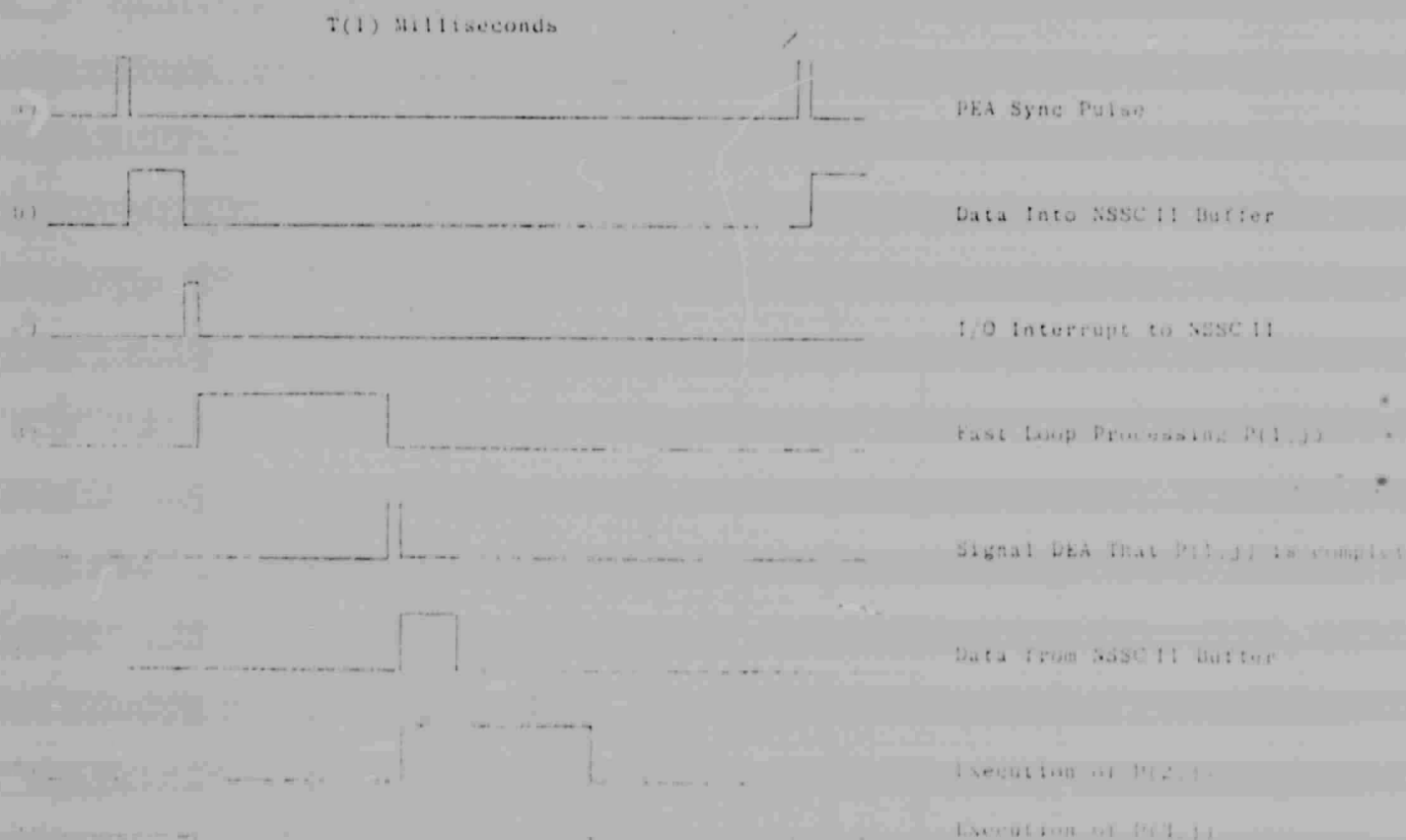


FIGURE 3



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